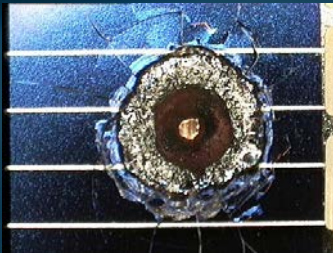


Advanced Life Support and In Situ Resource Utilisation –State of the Art in Europe

Robert Lindner/Christel Paillé,
ESA ESTEC TEC-MMG

- Life Support, Definition and why we need it
- Traditional Life Support
- Advanced Closed Loop Life Support Systems
- Current activities in ESA/Europe
- In Situ Resource Utilisation: Living off the land
- Application of the Resources
- ISRU processes, state of the art, advantages and disadvantages
- Current efforts in Europe/ISRU Demo Mission

Different environment...still "same" metabolism



7000 > 10cm

Daily Input* /pers.

O₂: 0.9 kg
 Water: 3.3 kg
 Food: 0.8 kg

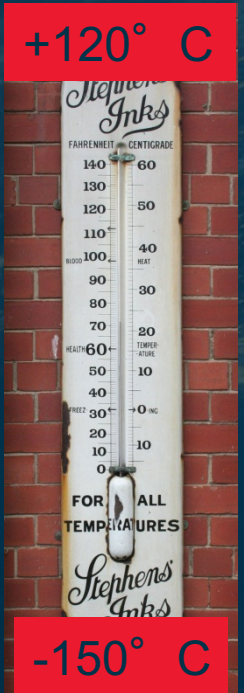
TOTAL :~5 kg



Daily Output* /pers.

CO₂: 1 kg
 Urine: 1.8 kg
 Perspiration: 1.8 kg
 Faeces: 0.4 kg
 Contaminants (chemical & microbial): traces

TOTAL :~5 kg



10⁻¹¹ hPa

Life Support System Functions

Life support systems for any crewed space flights include the following functions:

- Air revitalisation
- Potable Water supply
- Wastes management
- Food supply
- Consumables quality control
- Chemical and microbial contaminants control



Traditional Life Support Systems

Traditional Life Support Systems rely heavily on the logistics capabilities and depending on the space vehicle / space mission (Vostok, Salyut, MIR, Mercury Gemini, Apollo, Skylab, Shuttle, ISS):

- Carbon dioxide collected and/or vented
- Oxygen brought from Earth or produced via water electrolysis
- Food brought from Earth
- Water recycled to various levels
- Wastes collected and disposed by return cargo
- Microbial contaminants mostly analysed on ground
- Chemical contaminants

→ limited sustainability and overall resiliency



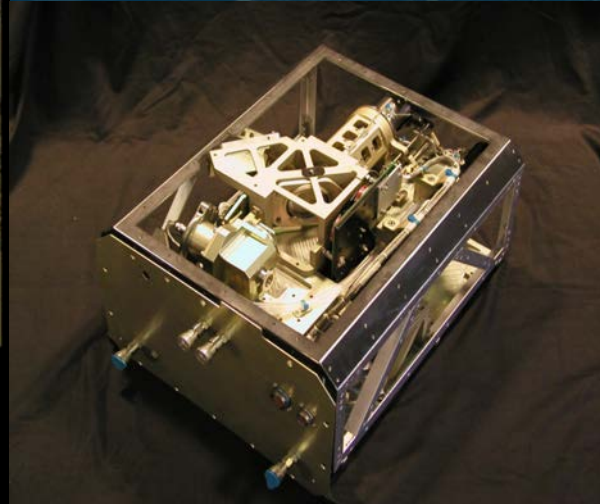
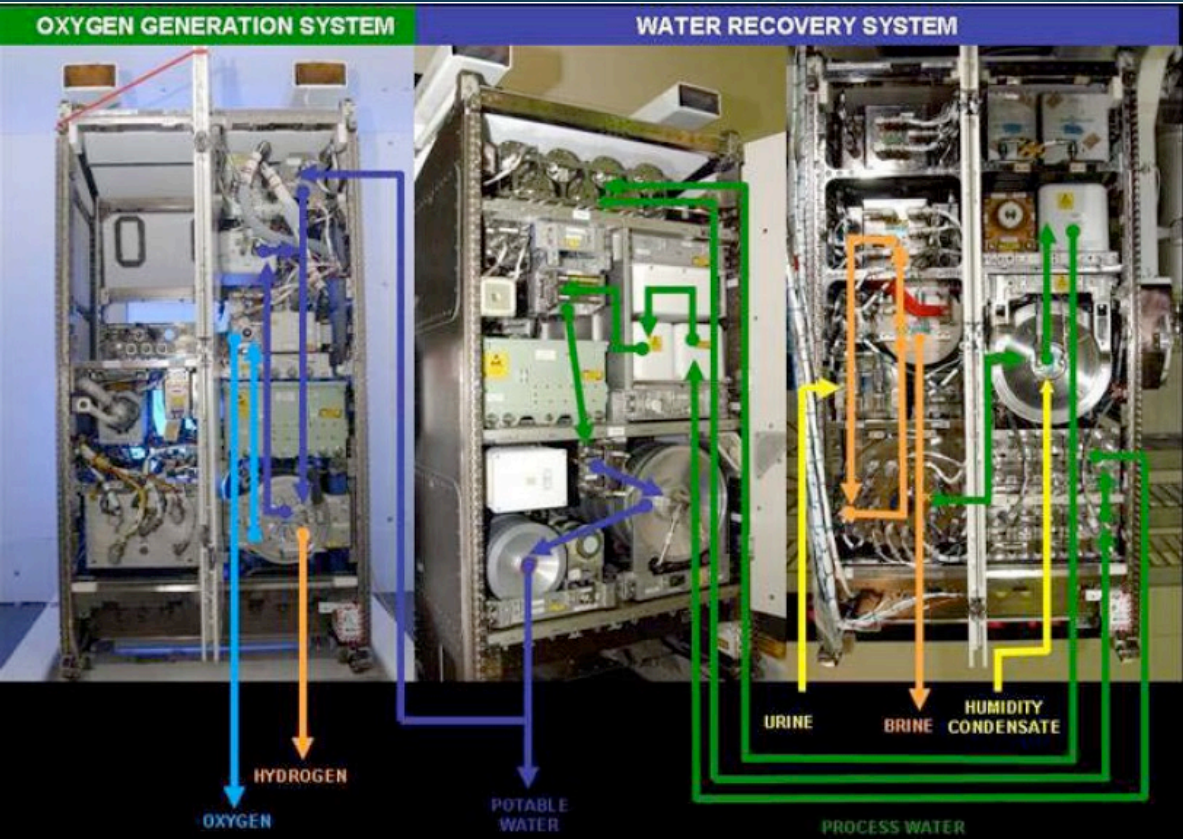
~400 Km



Advanced Life Support Systems

Life Support Systems evolves towards a closed system

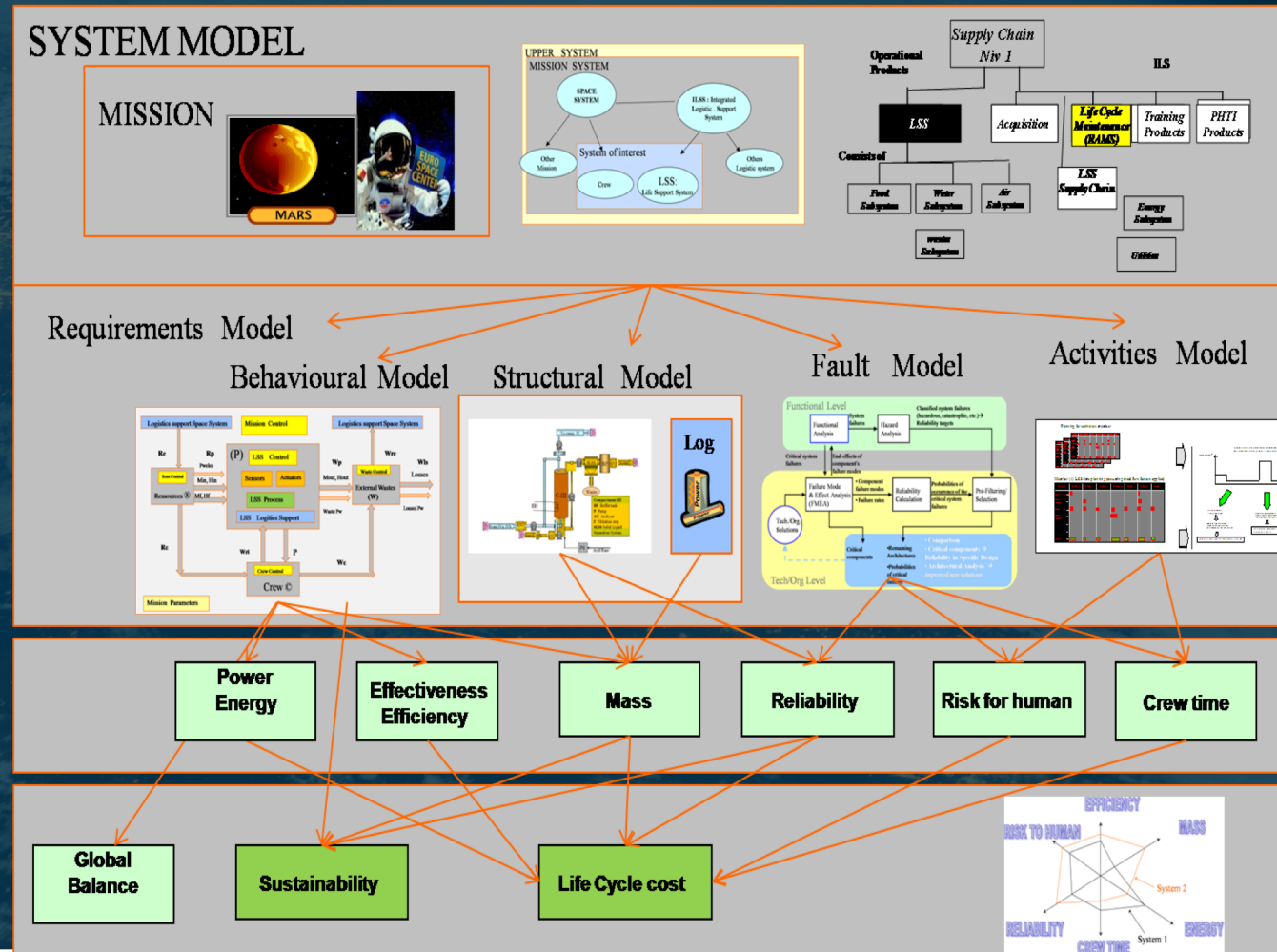
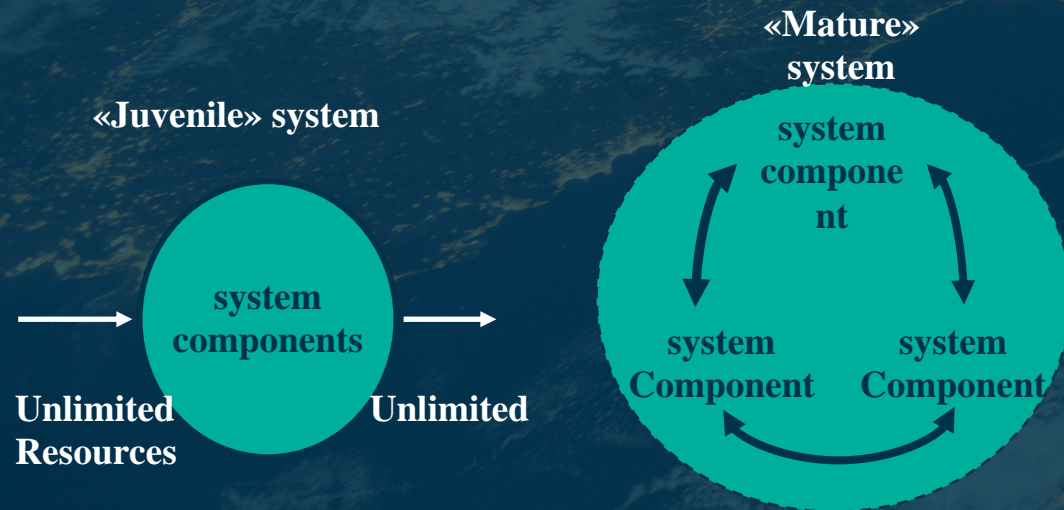
- Air revitalisation system targeting CO₂ processing into O₂ (Europe)
- Urine/condensate processing targeting water and oxygen production (US)



Current activities in Europe

The approach in Europe is strongly directed towards closed regenerative system

Challenge: assemble the recycling technologies to reach the highest level of closure and the best set of parameters such as mass, efficiency, energy, safety and crew time



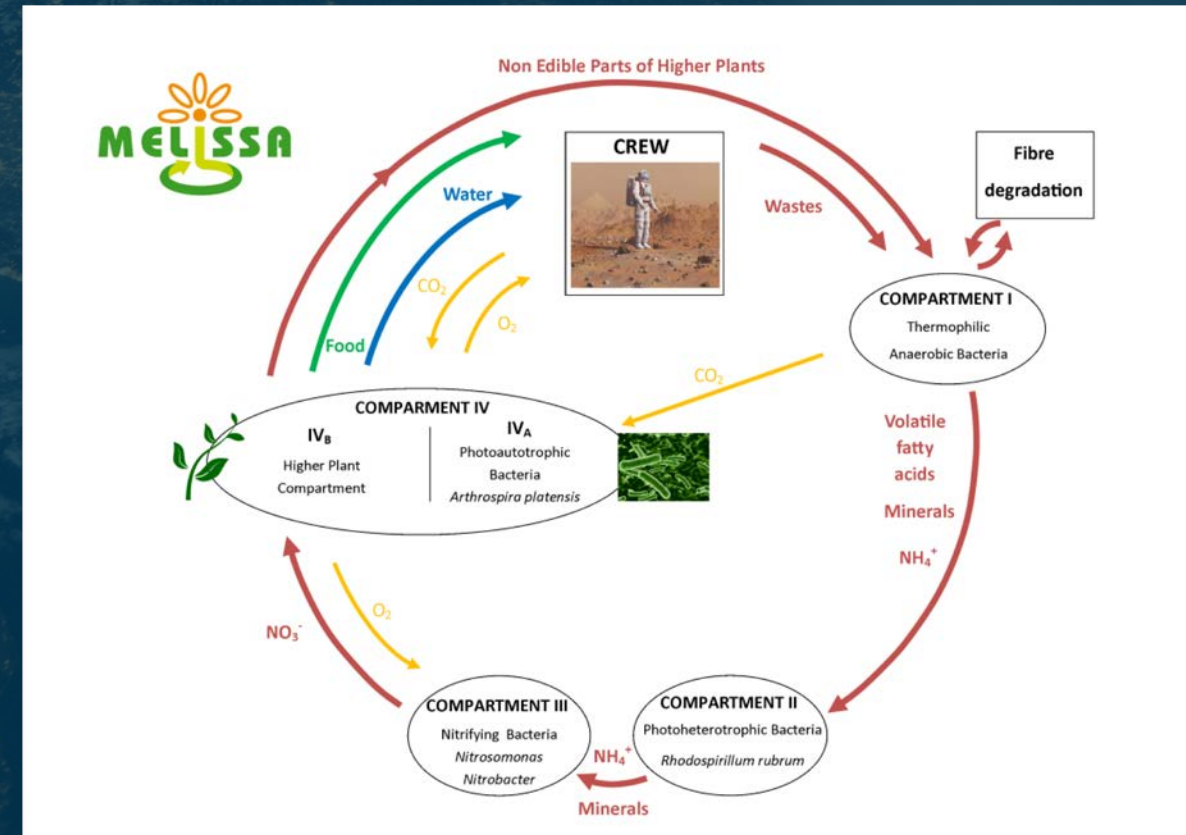
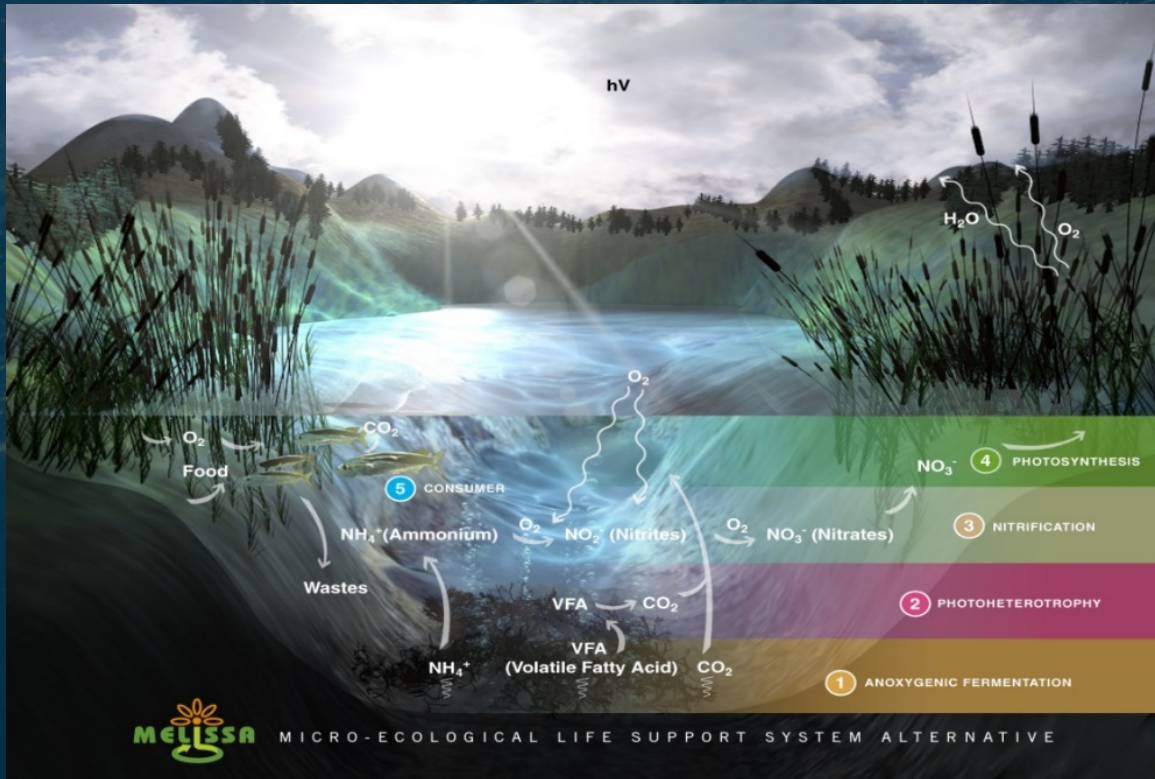
Current activities in Europe: MELiSSA project

Project initiated in 1989

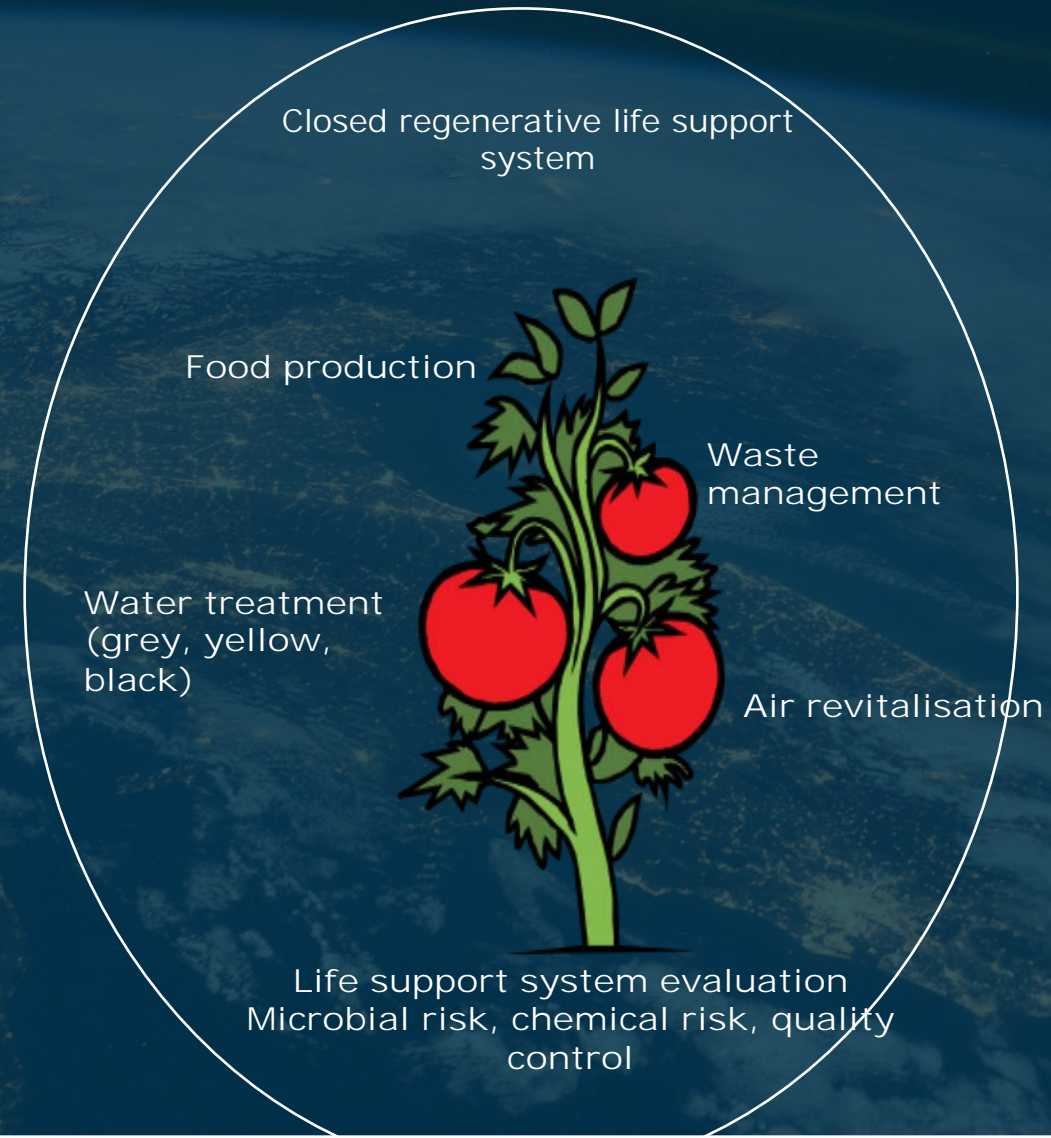
Governed by the MELiSSA Memorandum of Understanding (14 Partners)

More than 40 European organisations, over 12 countries

Approx. 100 scientists/engineers



- Phase 1: Basic R&D
 - System studies,
 - Sub-systems R&D,
- Phase 2: Preliminary Space experiments
 - Scientific data acquisition
 - Technology demonstration
- Phase 3: Ground demonstration/integration
- Education and communication



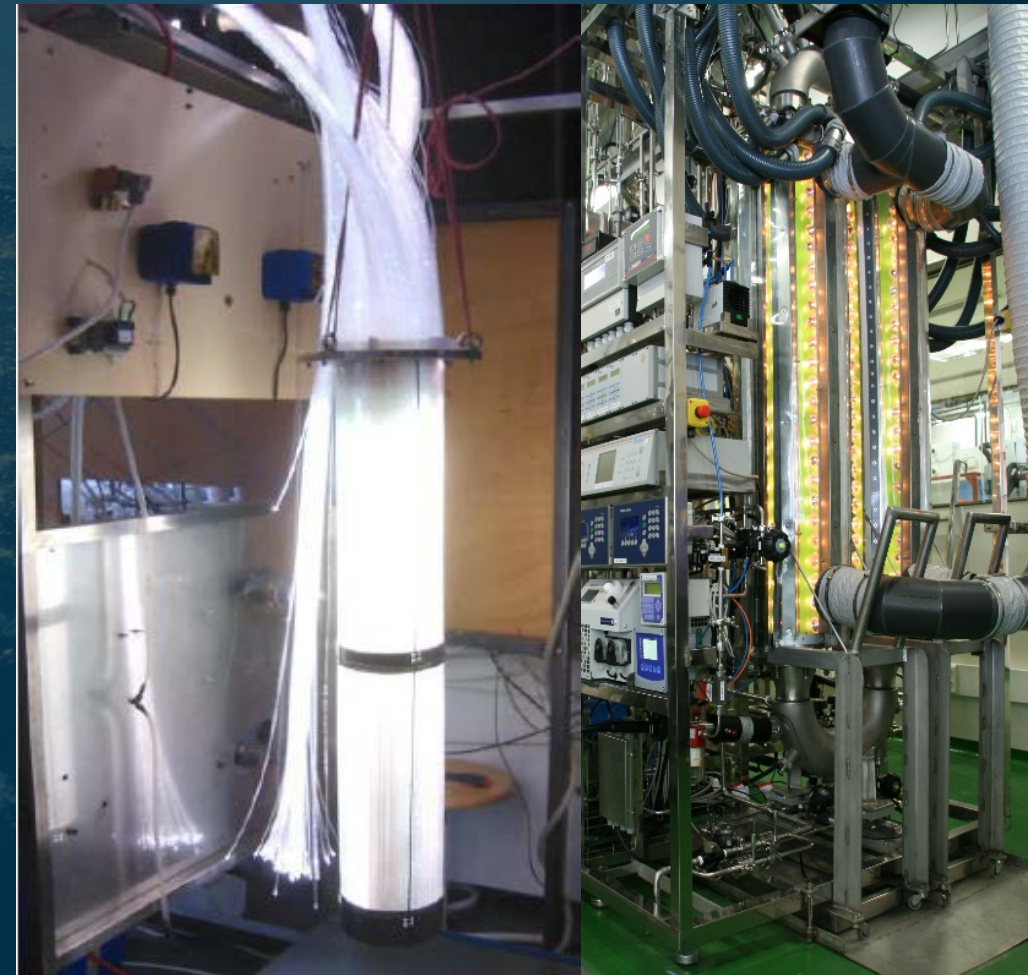
MELiSSA phase 1

Air revitalization function based on cyanobacteria photosynthetic process

Several size and oxygen production capabilities are available

Current work focuses on intensification of volumetric production

Process demonstrated in micro-gravity environment



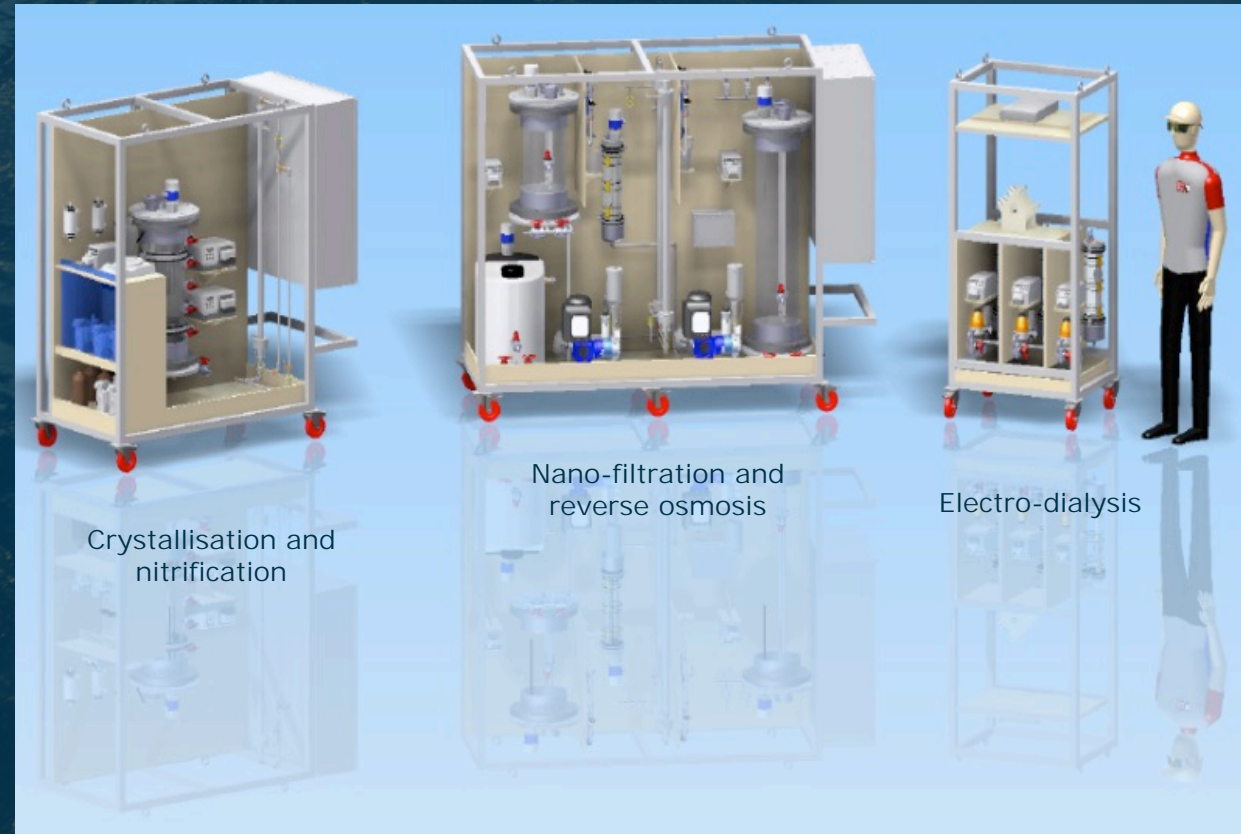
MELiSSA phase 1

Water treatment

Water represents the major mass of consumable to be uploaded and results in several streams, which need to be treated separately:

- grey water (condensate, shower,...)
- urine

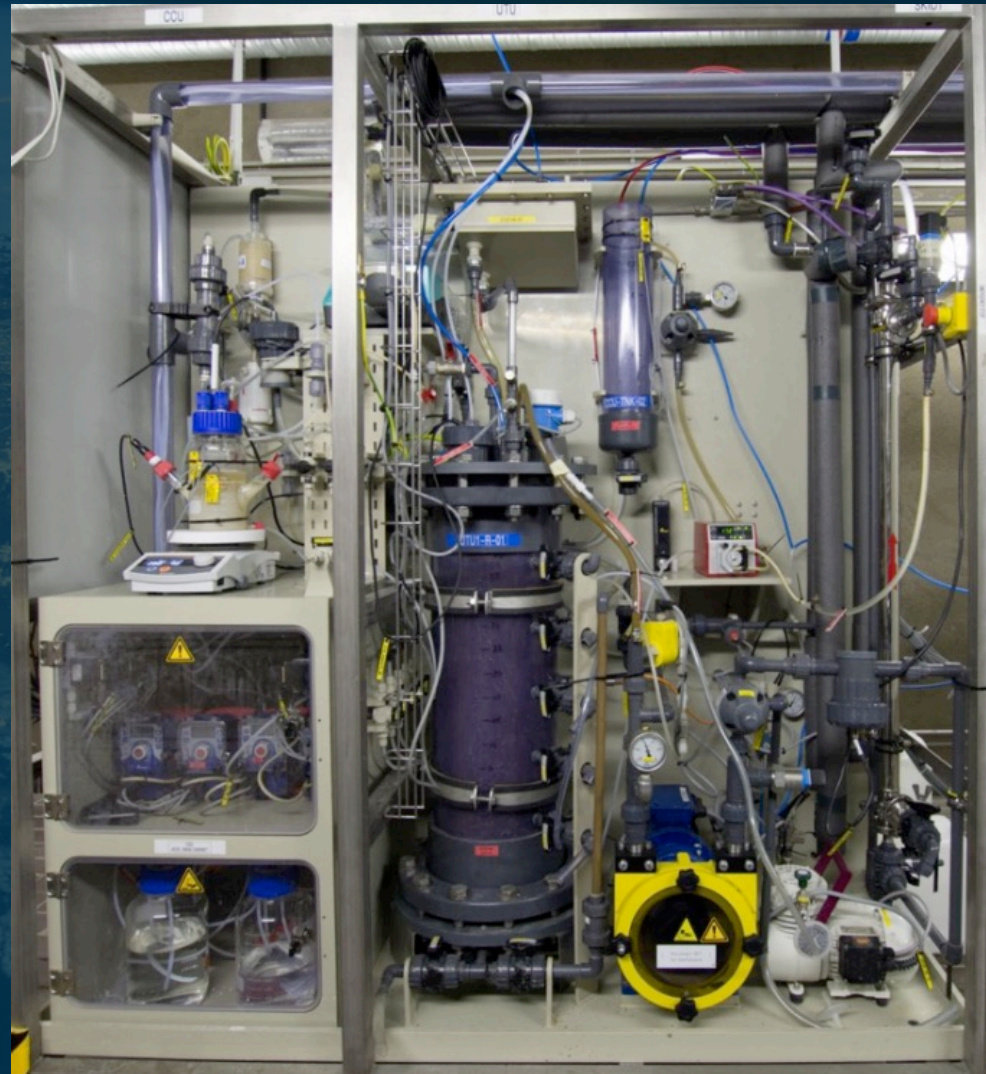
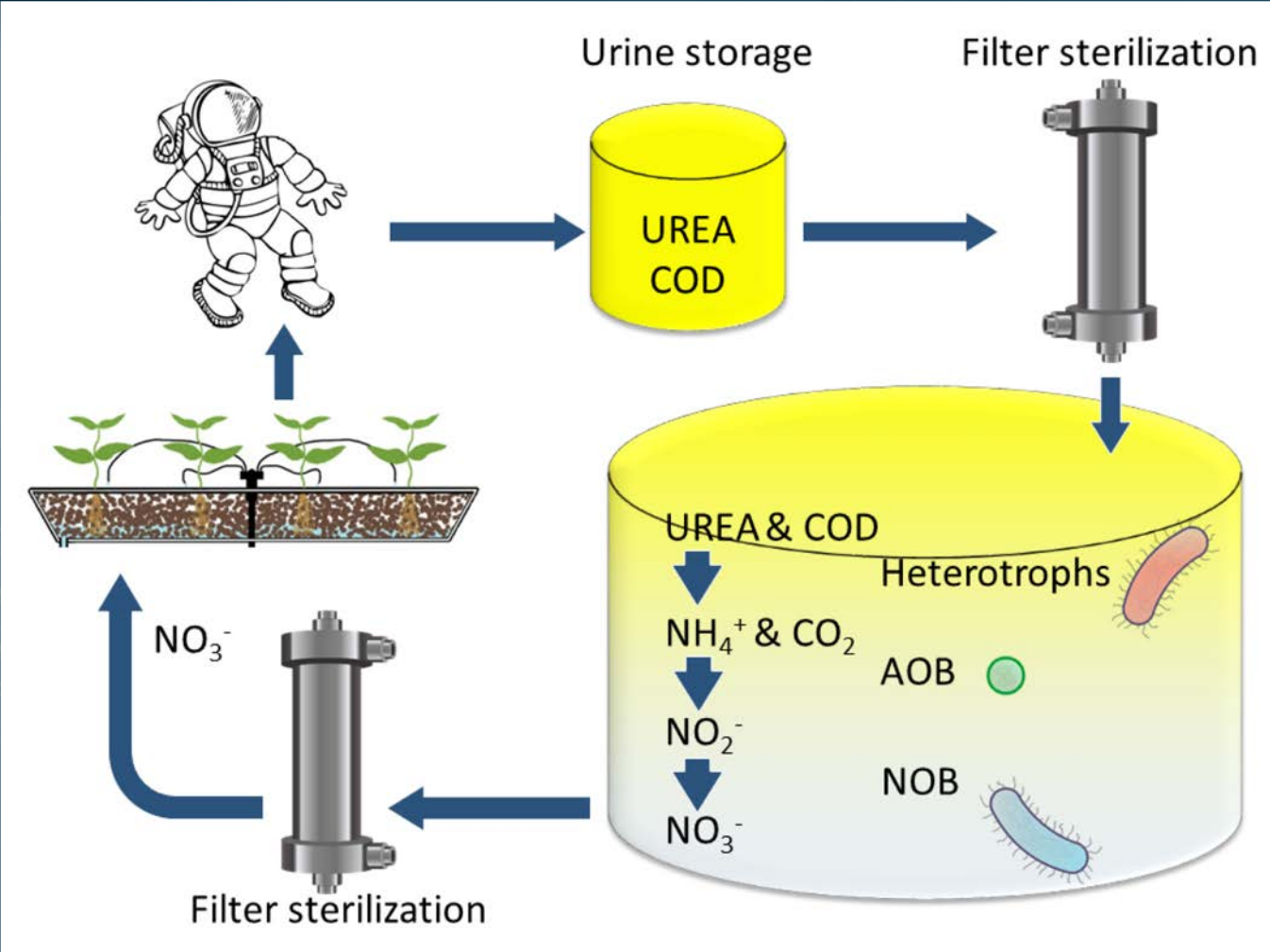
Prototype for space mission: up to 90% water recovery for treatment of urine and shower water produced by 1 person



MELiSSA phase 1



Urine conversion



Waste degradation

Waste are composed of:

- crew metabolic organic matter (fecal matter)

- toilet paper

- non-edible part of crops

Complementary techniques under investigations

Liquefying bioreactor

- microbial consortium metagenomics characterisation

- functional mathematical model

Fibre degradation:

- supercritical oxidation

- bio-anode (microbiological fuel cell)

MELiSSA phase 1

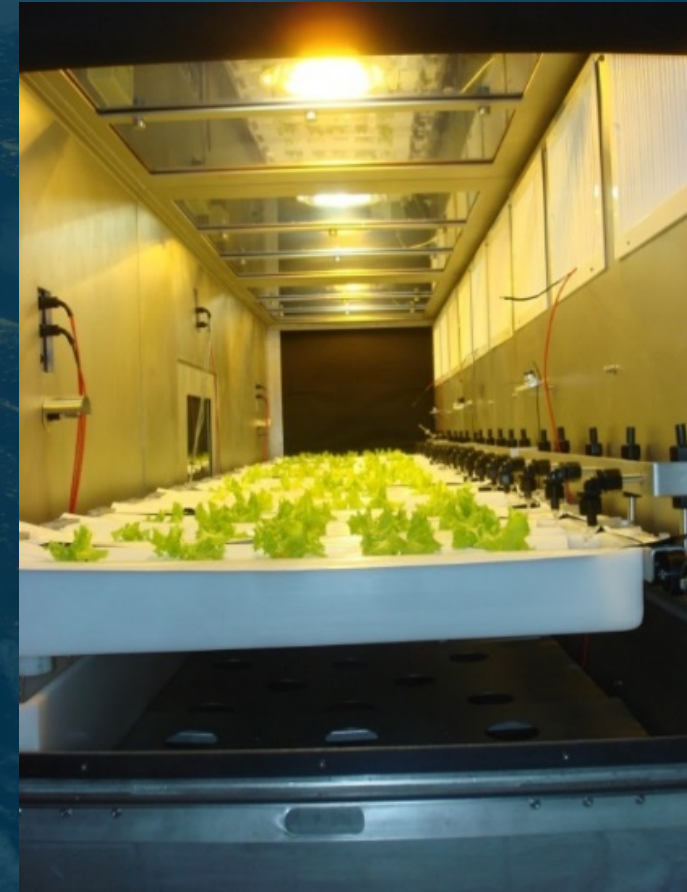
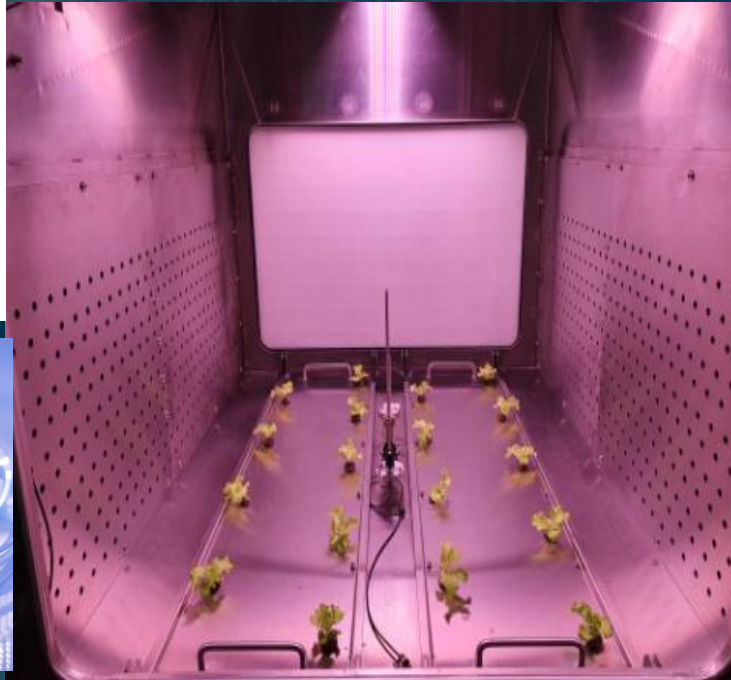
Food Production and preparation

Food (i.e. edible part of higher plants) is the resulting product of the following functions:

Tests on crops

Tests on recipes

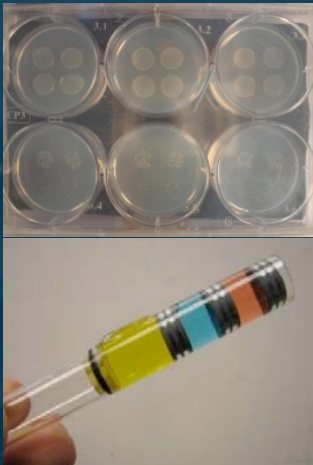
Tests with consumer



MELiSSA phase 2

From “test tubes” to bioreactor (or “small scale individual compartment”)

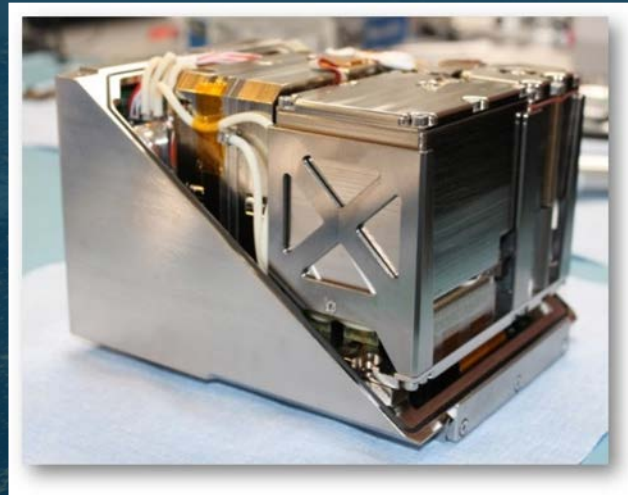
From individual compartment to connected compartments and technology demonstrators



MESSAGE 1
MESSAGE 2
BASE
NITRIMEL
BISTRO



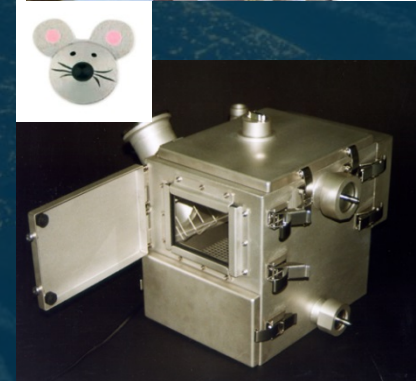
WAPS



ArtEMISS



URINIS



BIORAT 1

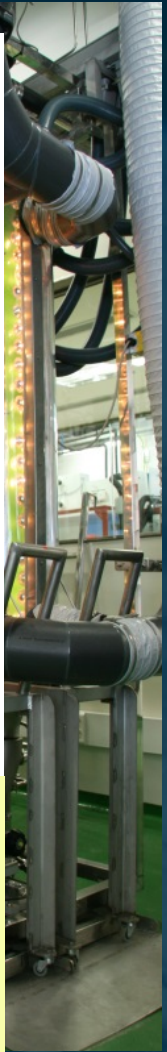
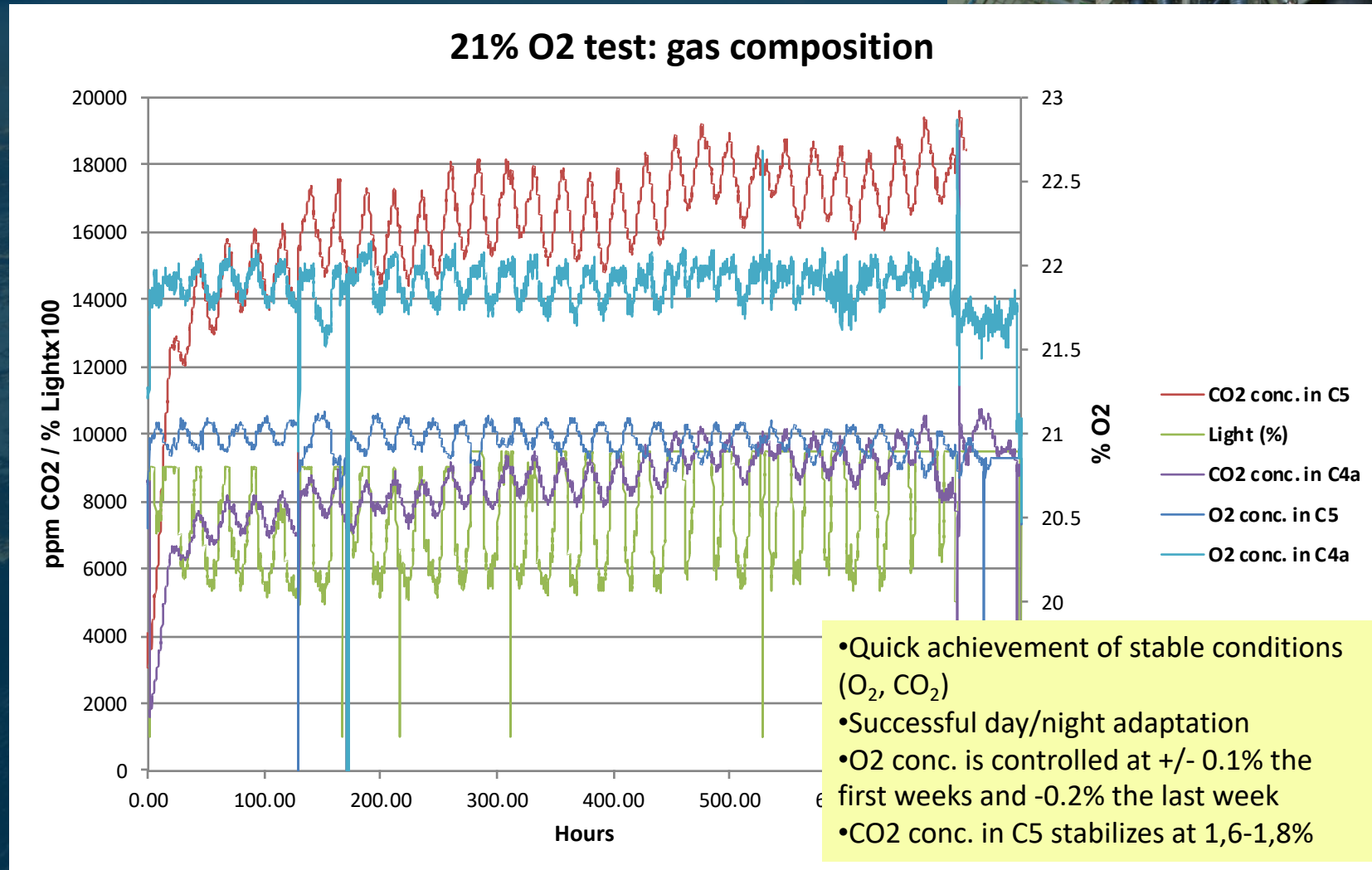
MELiSSA phase 3

MELiSSA Pilot Plant

Demonstration of air loop closure

Conversion CO₂/O₂

Dynamically controlled

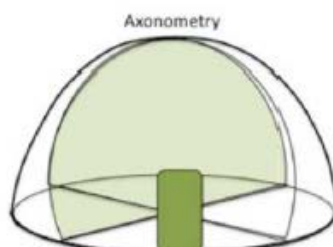


System studies


- Space Life support system preliminary sizing
- Moon greenhouse

PS selected concepts for further analysis

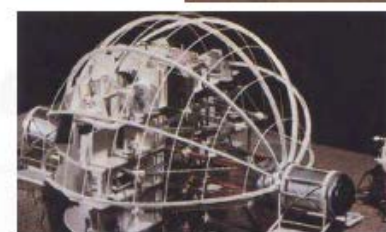
1 - INFLATABLE DOME - ONE MEMBRANE



Axonometry




SICSA MarsLab Concept [2004]




SICSA LunarHab Concept [1980]

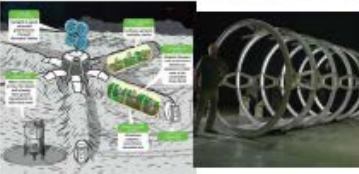
2 - INFLATABLE CYLINDER W. INT. STRUCTURE



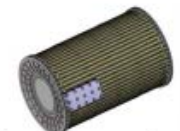
Section Axonometry



LGH Arizona University [on-going]

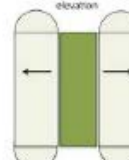


NASA/ILC Lunar Habitat [1996]




TASI/Aero Sekur STEPS2 [on-going]

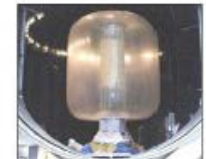
3 - INFLATABLE CYLINDER W. INT. RIGID CORE




Elevation



NASA/ILC Dover/TASI TransHab [2000]


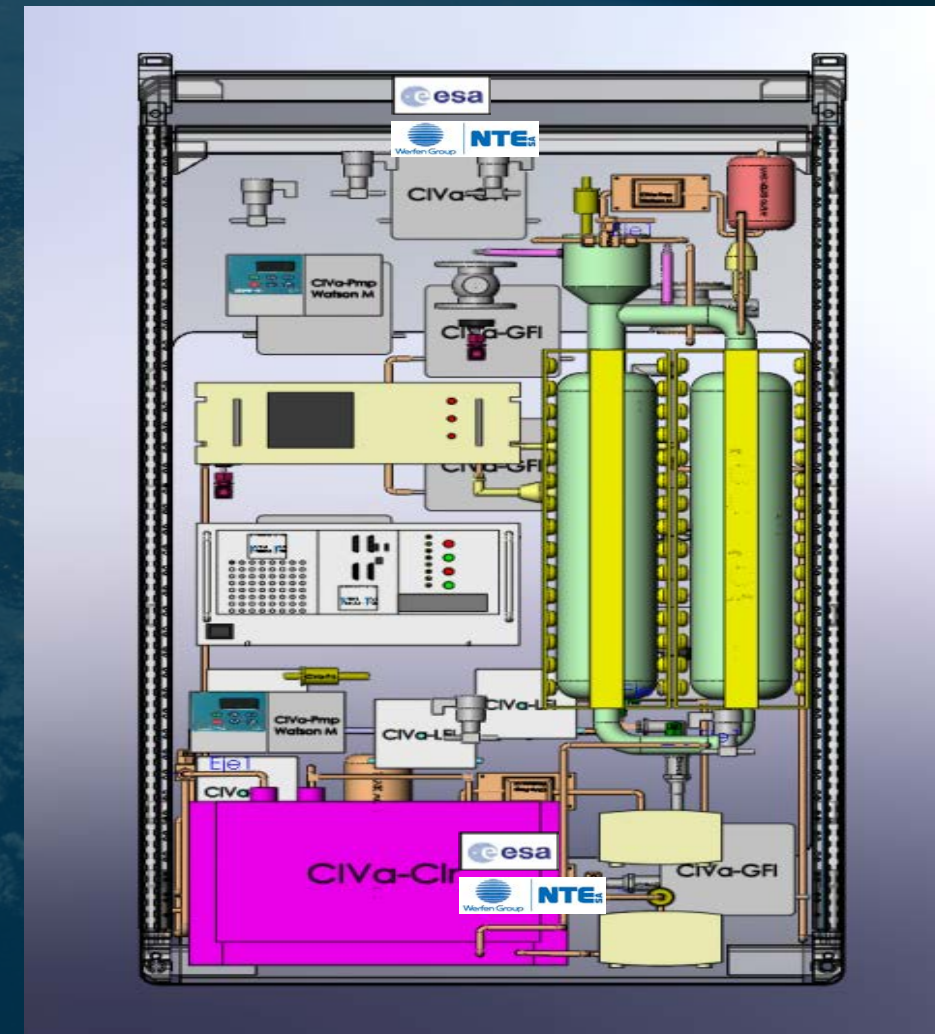


NASA/Bigelow Genesis I, II and BEAM [on-going]



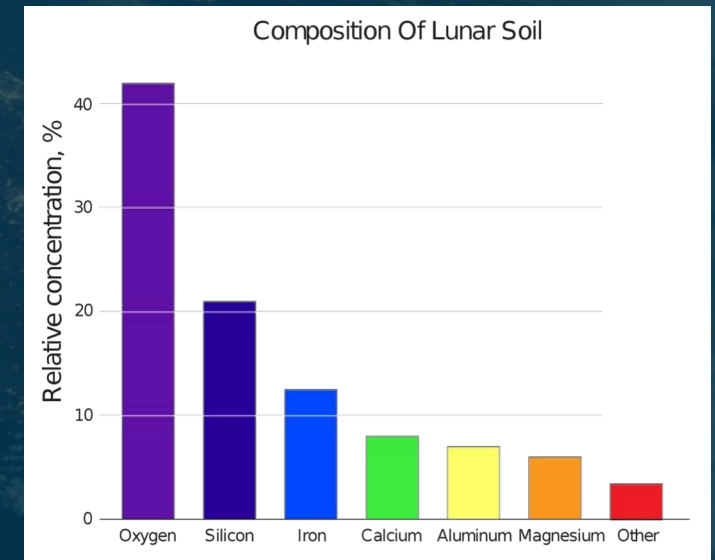
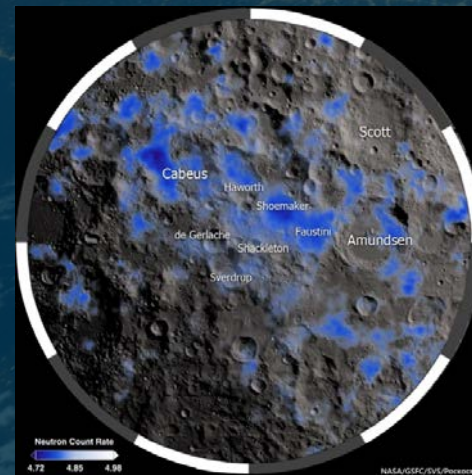
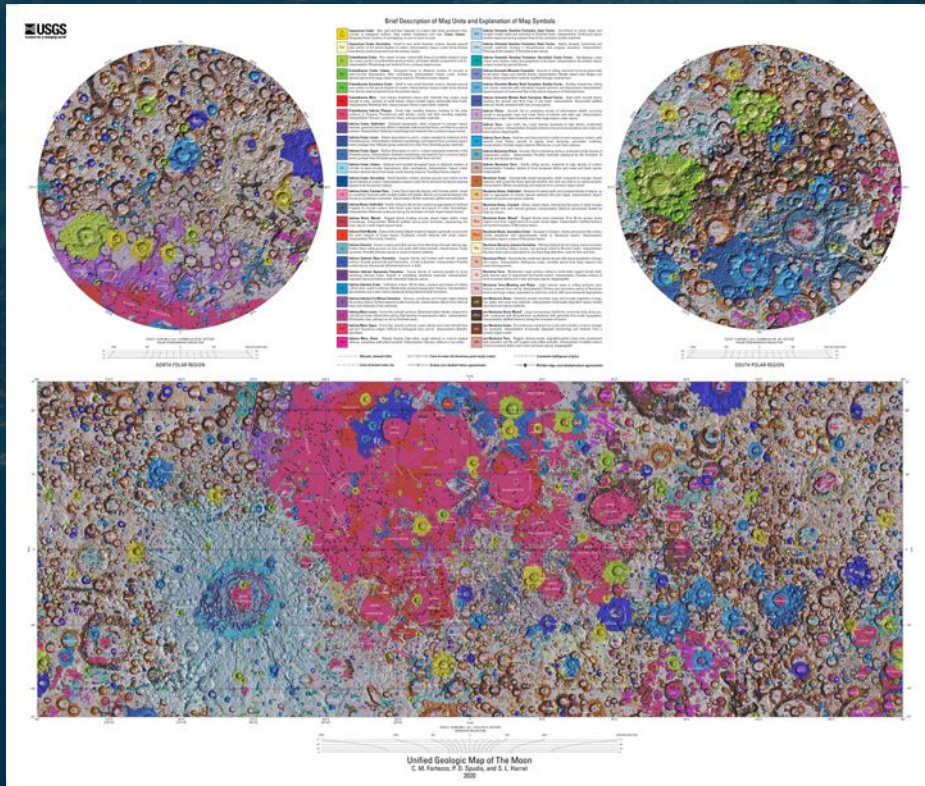
ESA/TASI/Aero Sekur IMOD [2006]

THALES ALENIA SPACE INTERNAL

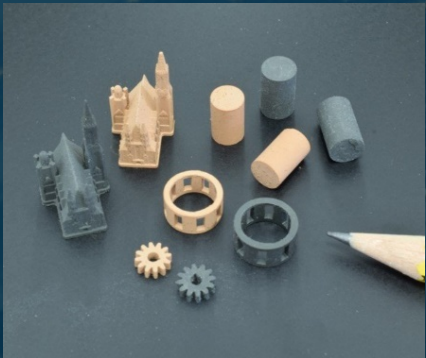
In Situ Resource Utilisation: Living off the land

- Exploration and permanent presence on planetary bodies requires best possible use of the resources
- Resources which can be used on a planetary body for sustainable presence
- Water and mineral oxygen of primary interest



Applications of the resources

- Oxygen and hydrogen as energy source: Propulsion, habitability
- For Life Support
- As construction and as building material (regolith)



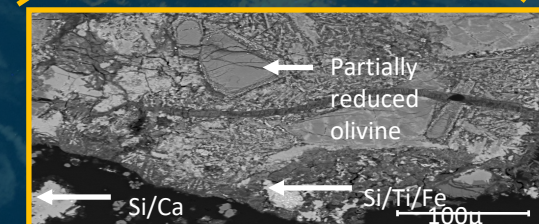
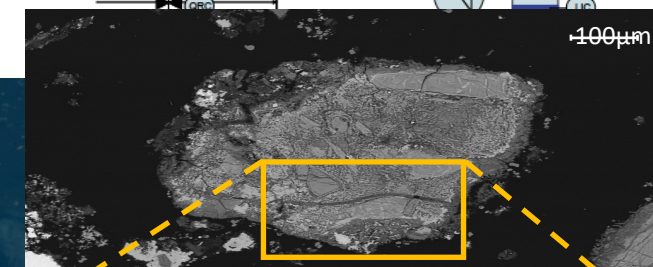
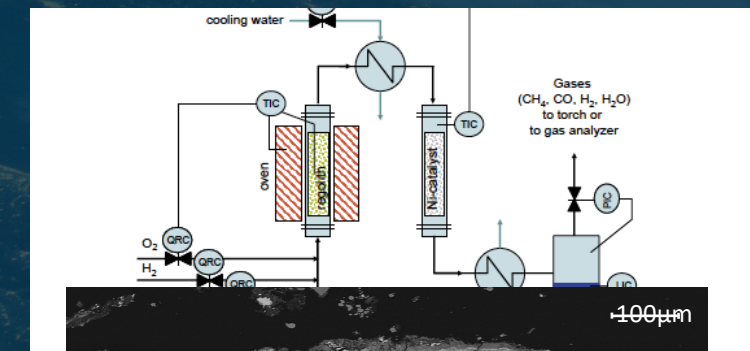
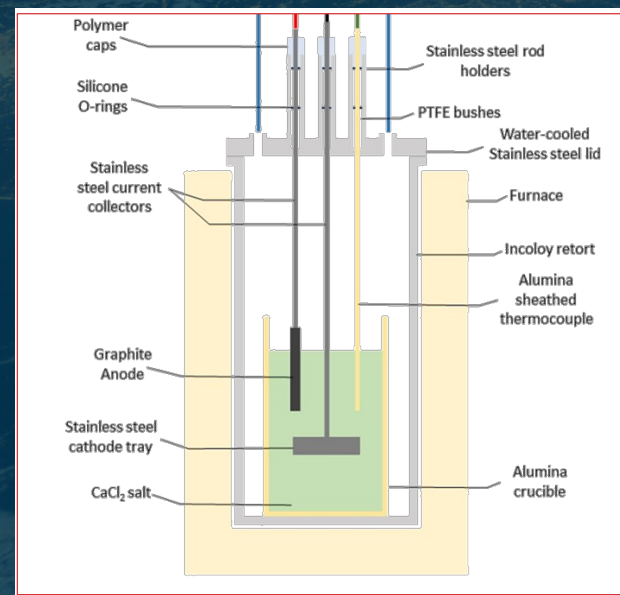
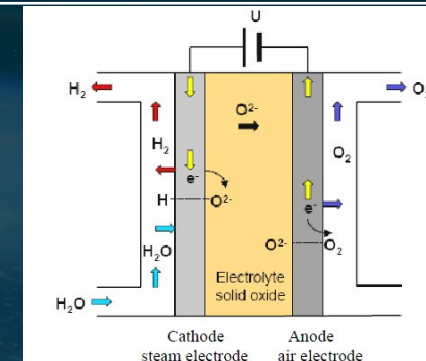
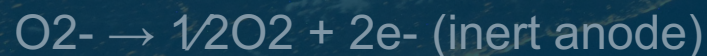
Development and qualification of processes for oxygen production

- Water Electrolysis
- Carbothermal reduction (1) $\text{CH}_4 + 0.5 \text{SiO}_2(\text{s}) \rightarrow 0.5 \text{Si}(\text{s}) + \text{CO} + 2 \text{H}_2$
- Methanation reaction: $3\text{H}_2 + \text{CO} \rightarrow \text{CH}_4 + \text{H}_2\text{O}$
- Hydrogen Reduction: $\text{FeTiO}_3 + \text{H}_2 \rightarrow \text{H}_2\text{O} + \text{TiO}_2 + \text{Fe}(\text{s})$
- Molten Salt Electrolysis:

AT THE CATHODE:



AT THE ANODE:

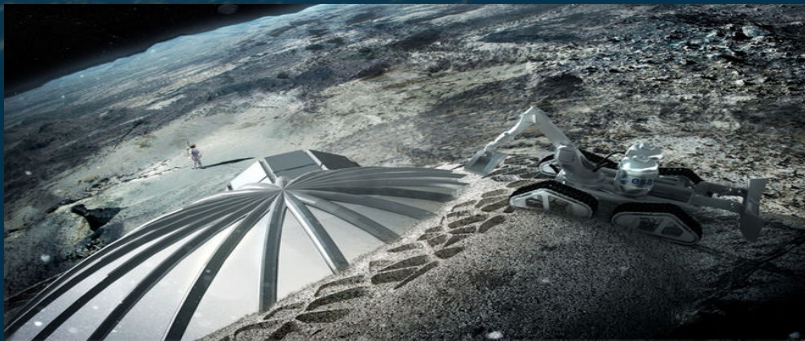


- Molten Regolith Electrolysis
- ...

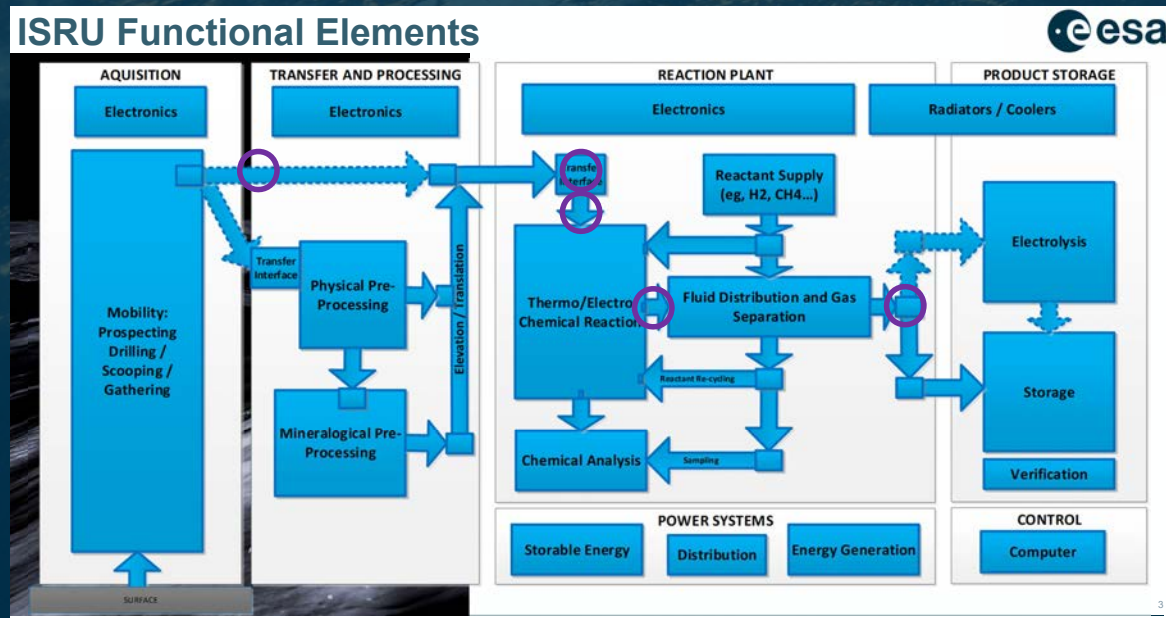
Advantages and disadvantages

- Water electrolysis: Simple and (relatively low) technological risk, Challenges is to get hold of the water (volatiles, PSR = -250C, access), provides oxygen (and hydrogen) directly
- Carbothermal reduction: Well known process, requires post processing (methanation), consumables form earth (CH4), low yield (5-10%), energy intensive (~900-1000C)
- Well known process, low yield (5-10%), specific to Ilmenite (FeTiO3), energy intensive
- Molten Salt Electrolysis (FFC Cambridge process): High Yield, (~70% oxygen), but technological risk and as well energy intensive (900C)

- Work on Ground Demonstrator for carbothermal/hydrogen reduction processes
- Work on industry side and internal R&D on molten salt electrolysis: Increase energy efficiency, corrosion, adaptation, materials to space environment...)
- Energy storage aspects (reversible solid oxide electrolysis)
- Phase A/B of an ISRU demonstrator mission to be scheduled for 2028
- Establishment of the European Resource Utilisation Centre, ESRIC (Luxemburg), bundling R&D and user applications (commercial, institutional) for resource utilization in space
- Work on additive and in-space manufacturing: Building materials, habitats, advanced engineering materials
- Supporting technology developments (dust resilience, efficient heating, reactor technology...)



- Objective is to produce oxygen on the surface of the moon and demonstrate the ISRU functional elements
- Phase A study finalized
- Phase B1 study to start in 2021: Challenge and then down selection of one concept
- Payload shall weigh no more than 80kg
- Launch with commercial or institutional lander
- Technology maturation necessary



Open Call on space resources: www.ideas.esa.int

More information on MELiSSA: www.melissafoundation.org

Thank you!

Robert.Lindner@esa.int

Christel.Paille@esa.int